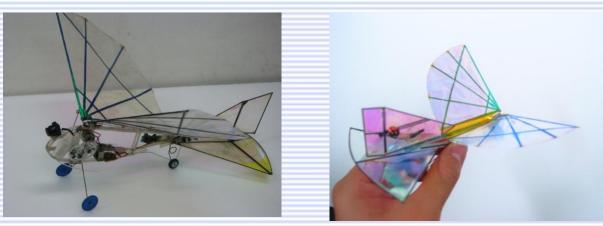


Flapping MAV



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## Contents



- Characteristics
  - Materials
  - Research on wing structure motivated by insect wing
- 36cm Flapping MAV

   capable of vision sensing and Take-off/Landing
- 28cm Flapping MAV
- 15cm Flapping MAV



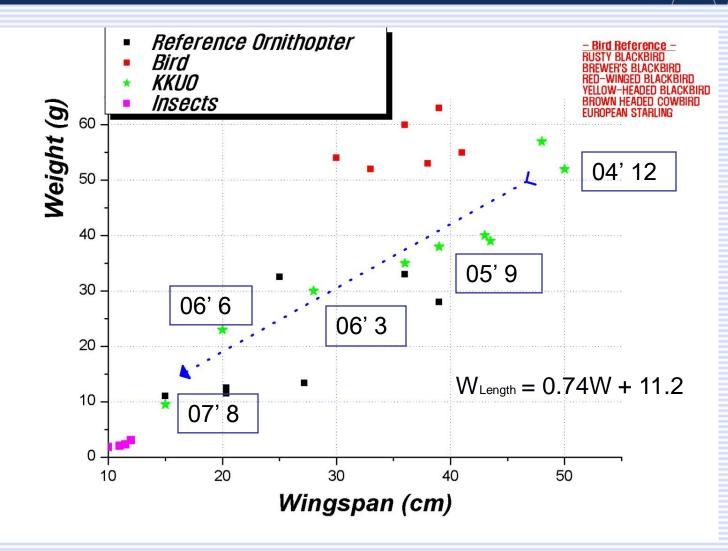


#### History

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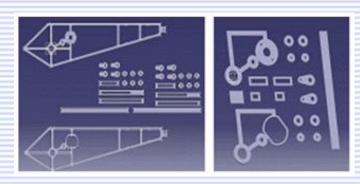




### Characteristics - materials

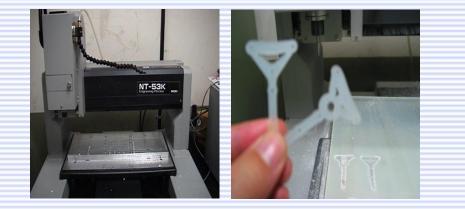


#### Designed using CATIA



Manufacture by CNC machine

- Materials
  - Body : glass/epoxy
  - Wing frame & shaft : carbon rod
  - Wing skin : epoxy film







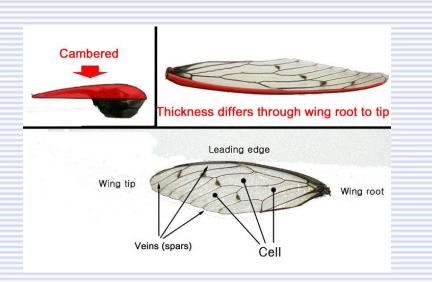




## Characteristics - wings

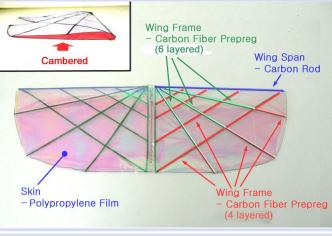


Insect wing(cicada) analysis •



- Characteristics of insect wing •
  - Cell-type wing
  - Camber wing on both spanwise and chord direction

**Bio-mimetic wing design** •

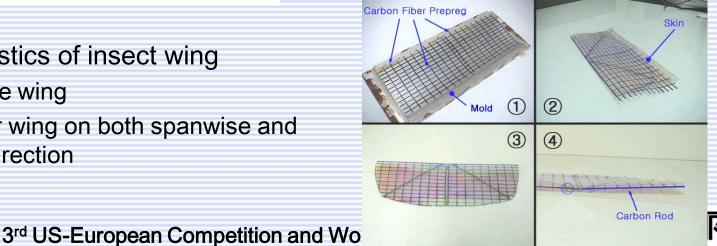


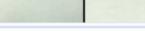
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**Procedures** •





## Characteristics - wings

#### Wing Shape Configuration

- The deformed shape of ornithopter wings during the flapping motion
- The deformation of wings of ornithopter and cicada

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 Different flapping shape due to the elasticity in chord direction









Landing gearCamera attached

Component Part	Mass	Product	Specification
Motor	6.09 g	B2C Motor	Input Voltage (V) 5
Battery	10.0 g	210mA	Output Voltage (V) 8.4
Speed Controller	1.22 g	Falcon HF ESC	
R/C Receiver	2.04 g	GWS Receiver	
Fuselage and Gear Box	6.44 g		Glass plate
Wing Structure	3.27 g		Carbon rod
Camera &transmitter	+6.05 g		
Total Mass	44.60 g(+6.05g)		



Take off

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Wing Span	36 cm	Gear Ratio	28:1 reduction
Wing Area	$432 \text{ cm}^2$	Frequency	20 Hz
Weight	50 g	Up Stroke °	35°
Wing loading	$0.115 \text{g/cm}^2$	Down Stroke °	0 °
Fuselage	23 cm	Flight Duration	15 min
3 <sup>rd</sup> US-European Competition and Workshop on MAV Systems			





Flapping and Gliding

Vision data from camera







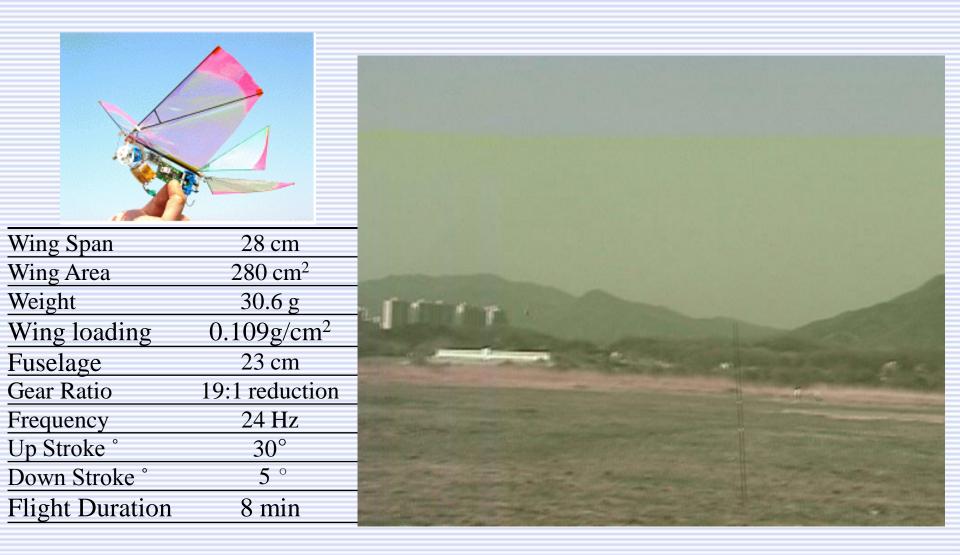


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Component Part	Mass	Product	Specification
Motor	6.09 g	B2C Motor	Input Voltage (V) 5
Battery	7.94 g	145mA battery	Output Voltage (V) 8.4
Speed Controller	1.22 g	Falcon HF ESC	
R/C Receiver	2.04 g	GWS Receiver	
Fuselage and Gear Box	6.44 g		Glass plate
Wing Structure	3.27 g		Carbon road
Total Mass	30.60 g(+3.05g)		
KU KONKUK       3rd US-European Competition and Workshop on MAV Systems       Smart Robot Competition and Workshop on MAV Systems			

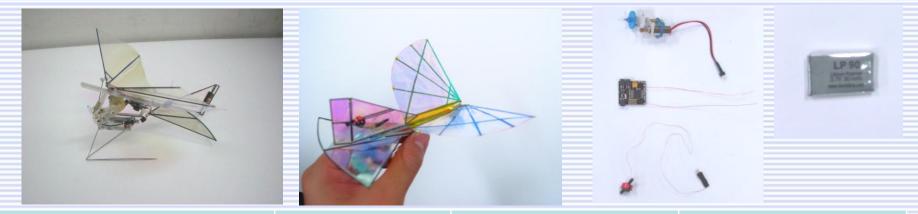










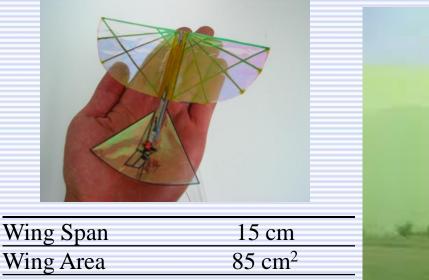


Component Part	Mass	Product	Specification
Motor	1.6 g	Pager motor	Input Voltage (V) 1~3
Battery	1.3 g	60mA battery	Output Voltage (V) 3.7
Speed Controller R/C Receiver	0.8 g		
Fuselage and Gear Box	0.8 g		Glass plate
Wing Structure	6.44 g		Carbon road
Total Mass	8.7 g		









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Weight	8.7 g
Wing loading	$0.102 \text{ g/cm}^2$
Fuselage	15 cm
Gear Ratio	16:1 reduction
Frequency	30 Hz
Up Stroke °	$40^{\circ}$
Down Stroke °	5 °
Flight Duration	1 min







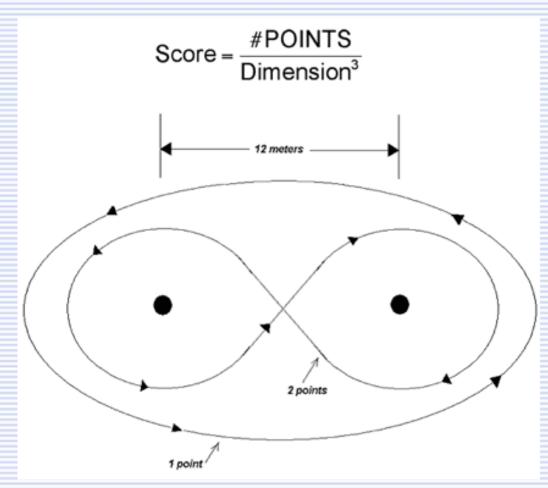


# Thank you …





### Ornithopter mission



KONKUK UNIVERSITY Ranked 4<sup>th</sup> Place at 9<sup>th</sup> Int. MAV Competition 2005. 6. Seoul Korea

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Ranked 2<sup>nd</sup> Place at 10<sup>th</sup> Int. MAV Competition 2006. 5. BYU Utah USA



#### Component Specification



Speed Controller		Falcon HF ESC for 2 Li-poly cellsThe Falcon High Frequency 100kHz 2 cell ESC isdesigned to work with 2 Li-poly cells. The ESC incorporates a 5vBEC to power the radio.Data: 1.45g complete with all onnectors and wire.Dimensions: 11mmx15mmx3mmBEC: 5v easily handles 3-4 servos.
Receiver	Contraction of the second seco	Gws Rx with JST plugsGWS Rx with JST 4.8 - 6 VDimensions: 15 x 25 x 10mmRange approx. 150m4.3g with micro crystal3.8g without housing
Actuators		Falcon 1.7g Servo         Falcon Coreless digital 1.7g servo with JST connectors, Torque 25+grams         Operating voltage 3.3 - 5v         Length



3<sup>rd</sup> US-European Competition and Workshop on MAV Systems

KU KONKUK UNIVERSITY



#### Conceptual Design

Component Mass

3.6
Mass
6.09 g
7.94 g
1.22 g
2.04 g
6.44 g
3.27 g
30.60 g

- Maximum assuming weight : 35g
  - Require Wingspan Length = 35cm

#### Design Requirement

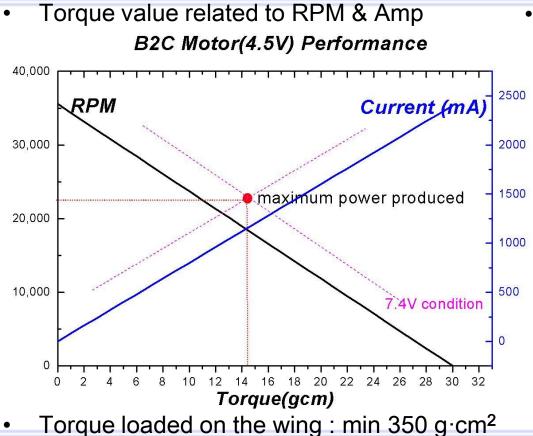
•

- Wing Span : 35cm
- Weight : max 35g
- Flight endurance : 15 min.
- Remote control range : max 100m
- Materials : composite materials
- Propulsion : electric motor
- Flight control : rudder, elevator



 $W_{Length} = 0.74W + 11.2$ 

#### Power System



- Selection of Motor :
  - B2C Motor



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Norminal Voltage (V)	4.5
Weight (g)	5.6
RPM (No load)	25500
Stall Torque (g•cm)	130

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- Maximum torque produced : 15 gcm
- Gear reduction ratio : 23:1

#### Gearbox System Design

- Type of Gear box
   Disadvantage
  - Heavier than other types
  - Non symmetric wing flapping motion
  - Structural weakness

 generates higher body vibration relatively A A

B

#### Advantage

 appropriate to transfer large force

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- Lighter than others
- Easy to repair

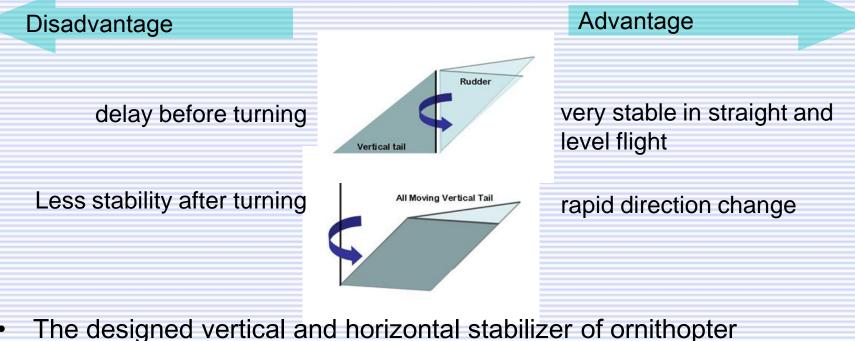
Strong structure

Type C was compatible in both weight and performance

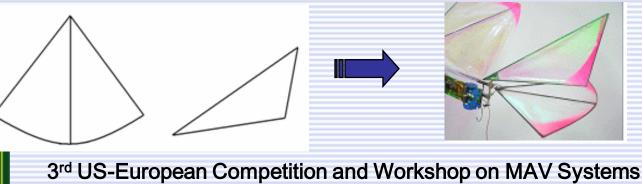




- **Tail System Design** •
- 2 types of vertical and horizontal stabilizer •



The designed vertical and horizontal stabilizer of ornithopter •



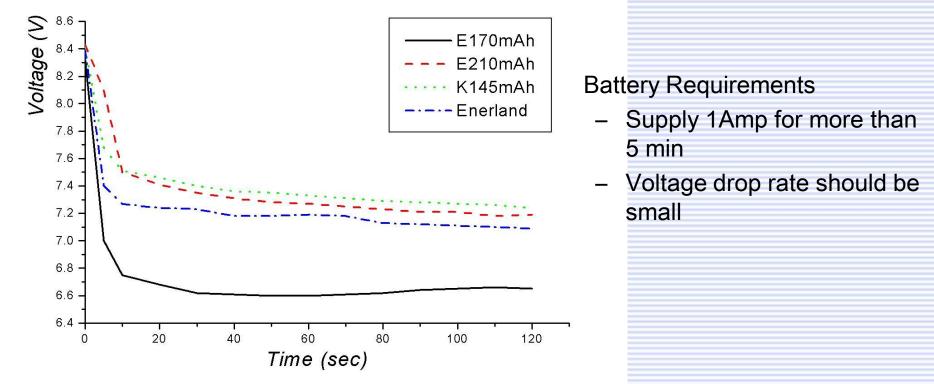




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- Battery Configuration
- Battery discharge performance
  - Battery type : Li-polymer battery (2-cell)
  - Measured with "Micro-Meter Model 100" device





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